

Deutscher Hängegleiterverband e.V. im DAeC Akkreditierte Musterprüfstelle für Hängegleiter und Gleitsegel nach DIN EN ISO/IEC 17020:2012-07

GS TESTFLUG LTF 2009 MAC PARA CHARGER 25

Test No 027416-GSTF09-786-Beni

Test date 07.08.2015

Location Garmisch /Osterfelder

Type Mac Para Charger 25

Test type GS Testflug LTF 2009

Test order Auftrag GS Musterprüfung Mac Para Charger 25 (MAC Para

Technology)

Customer MAC Para Technology

Test standard LTF NFL II-91/09 und NfL 2-60-14

Test standard 2 EN 926-2:2014

Expert Stocker

Result positive

Billing to: 100%

Technical peculiarities

13 Much

Datum / Unterschrift (Beni Stocker)

RESULTS PG test flight (general) Take off weight [kg] 75 Weight limit for certification [kg] 75 Number of pilots 1 test pilot Beni Stocker Harness type SUP'AIR 02 Harness category GH Minimum speed [km/h] 21 Trim speed [km/h] 32 Accelerated speed [km/h] 0 Accelerator used? Yes en : Klassifizierung en : Klassifizierung B EN: ERGEBNISDETAILS NACH LTF 2009 1 Inflation/take-off Rising behaviour Smooth, easy and constant rising Special take off technique required No 2 Landing Special landing technique required No 3 Speeds in straight flight Trim speed more than 30 km/h Yes Speed range using the controls larger than Yes 10 km/h Minimum speed Less than 25 km/h 4 Control movement Symmetric control pressure Increasing Symmetric control travel Greater than 55 cm 5 Pitch stability exiting accelerated flight Dive forward angle on exit Dive forward less than 30° Collapse occurs No 6 Pitch stability operating controls during accelerated flight Collapse occurs No 7 Roll stability and damping Oscillations Reducing

8 Stability in gentle spirals		Α
Tendency to return to straight flight	Spontaneous exit	
9 Behaviour in a steeply banked turn		Α
Sink rate after two turns	Up to 12 m/s	
10.1 Symmetric front collapse		Α
	Rocking back less than 45°	
	Spontaneous in less than 3 s	
Dive forward angle on exit		
	Entering a turn of less than 90°	
Cascade occurs	No	
10.2 Symmetric front collapse in acceler		В
	Rocking back less than 45°	
-	Spontaneous in 3 s to 5 s	
Dive forward angle on exit		
_	Entering a turn of less than 90°	
Cascade occurs	No	
11 Exiting deep stall (parachutal stall)		Α
Deep stall achieved		
	Spontaneous in less than 3 s	
Dive forward angle on exit		
<u> </u>	Changing course less than 45°	
Cascade occurs	No	
12 High angle of attack recovery		Α
Recovery	Spontaneous in less than 3 s	
Cascade occurs	No	
13 Recovery from a developed full stall		Α
Dive forward angle on exit		APPROXICATION TO THE TOTAL SERVICE SER
	No collapse	
Cascade occurs (other than collapses)	No	
Rocking back	Less than 45°	
Line tension	Most lines tight	
14.1 Asymmetric collapse 45-50%		A
Change of course until re-inflation	Less than 90°	
Maximum dive forward or roll angle		
Re-inflation behaviour	Spontaneous re-inflation	
Total change of course	Less than 360°	
Collapse on the opposite side occurs	No	
Twist occurs	110	
Cascade occurs	a No	
14.2 Asymmetric collapse 70-75%		В
Change of course until re-inflation		
Maximum dive forward or roll angle	=	
	Spontaneous re-inflation	
Total change of course		
Collapse on the opposite side occurs		
Twist occurs Cascade occurs		
14.3 Asymmetric collapse 45-50% in ac Change of course until re-inflation		Α
Maximum dive forward or roll angle		
	Spontaneous re-inflation	
Total change of course	•	
Collapse on the opposite side occurs		
Twist occurs		
Cascade occurs	s No	
14.4 Asymmetric collapse 70-75% in ac	celerated flight	В
Change of course until re-inflation		
Maximum dive forward or roll angle		
	r Spontaneous re-inflation	
Total change of course	Less than 360°	
Collapse on the opposite side occurs		
Twist occurs		
Cascade occurs	s No	

15 Directional control with a maintained	asymmetric collapse	А
Able to keep course	Yes	
180° turn away from the collapsed side possible in 10 s	Yes	
Amount of control range between turn and stall or spin	More than 50 % of the symmetric control travel	
16 Trim speed spin tendency		Α
Spin occurs	No	
17 Low speed spin tendency		Α
Spin occurs	No	
18 Recovery from a developed spin		Α
Spin rotation angle after release	Stons spinning in less than 90°	
Cascade occurs		
0.000.00		
19 B-line stall		Α
Change of course before release	Changing course less than 45°	
	Remains stable with straight span	
	Spontaneous in less than 3 s	
Dive forward angle on exit	•	
Cascade occurs		
Cascade occars		
20 Big ears		A
L	Dedicated controls	
Behaviour during big ears		
Recovery	Spontaneous in less than 3 s	
Dive forward angle on exit	Dive forward 0° to 30°	
21 Big ears in accelerated flight		Α
L.,	Dadicated controls	
	Dedicated controls	
Behaviour during big ears	Spontaneous in less than 3 s	
Dive forward angle on exit	·	
Behaviour immediately after releasing the		
accelerator while maintaining big ears		
22 Behaviour exiting a steep spiral		A
Tendency to return to straight flight	Spontaneous exit	
,	Less than 720°, spontaneous recovery	
Sink rate when evaluating spiral stability	14	
[m/s]		
23 Alternative means of directional con	trol	Α
180° turn achievable in 20 s	Yes	
Stall or spin occurs		
24 Any other flight procedure and/or c	onfiguration described in the	
user's manual	No other flight procedure or configuration described in the user's	
	manual	